MEASUREMENT OF RESIDUAL STRESSES IN STRUCTURAL ELEMENTS OF SHIPS USING SECTIONAL METHOD

ISMAEL SADOK, BSc. MSc., Ph.D., NAWS
Faculty of Engineering, Port Said, Suez Canal University

Abstract

The structural elements of ships are subjected to several manufacturing processes during the stages of production. As a result of these processes, residual stresses are created in the structural elements to a large extent.

Many techniques have been proposed and used for measuring residual stresses in metals. These techniques are classified into four groups: (a) stress-relaxation techniques, (b) X-ray diffraction techniques, (c) techniques using stress-sensitive properties, and (d) cracking techniques.

The main aim of this research is the measurement of residual stresses in structural elements of ships using sectional method. A digital electronic automation machine is used for marking the gage length and then measuring it before the processing and after the sectioning.

Comparison between the results of sectional method and the blind-hole drilling method has been made from accuracy and economical points of view.

It is concluded that the sectional method is adequate, accurate, and economical for residual stresses measurements in structural elements of ships.

1. Introduction

Residual stresses in a structural material component are those stresses which exist in the object without the application of any service or other external loads.

Manufacturing processes are the most common causes of residual stresses. In shipbuilding those stresses due to the cutting, forming, welding, subassembly, assembly, and heat treatment operations.

In many cases, residual stresses are the predominant factor contributing to structural failures and fatigue, mostly when an alternating service stress adds to the already present residual stress.
Many techniques have been proposed and used for measuring residual stresses in metals. In this research, the sectional method is used in order to measure residual stresses in structural elements of ship.

Experimental work has been done in the Laboratory of Strength of Material, Faculty of Mechanical Engineering and Naval Architecture, Zagreb, Yugoslavia.

2. SECTIONS AND MANUFACTURING PROCESSES USED IN SHIPBUILDING

There are several shapes and types of sections which are used in shipbuilding. Fig. 1 illustrates some of these sections used in present day construction. The sections range from simple flat bar, edge-welded to the supported plating, to a flanged plate girder with additional face bar reinforcement. Girders are also built up by welding a flange to a lighter web, larger angle sizes may be obtained by cutting one flange from a channel section.

![Diagram showing structural shapes](image)

**Fig. 1. Structural shapes**

Sections are stored in separate stock yards and fed into their individual shot-blasting and priming machines. The sections are cleaned by abrasive shot and then coated with a suitable prefabrication priming paint to a limited thickness for ease of welding. The areas of steel are therefore protected from corrosion during the various manufacturing processes which follow. The sections follow their individual paths to the marking or direct cutting machinery which produces the suitable dimensioned items. Edge preparation for welding may also be done. Various shaping operations now take place using cold frame benders. The sections transfer before, during and after the various processes in shipbuilding utilizing many handling appliances. The various steel parts in plate and section form are now joined together by welding to produce subassemblies, assemblies and units.
3. MEASUREMENT OF RESIDUAL STRESSES BY SECTIONING METHOD

The sectioning method is based on the principle that internal stresses are relieved by cutting the specimen into many strips of small cross section. The stresses distribution over a cross section can be determined with reasonable accuracy by measuring change in length of each strip and by applying Hooke's law.

The steps in the sectioning and slicing process are illustrated schematically in Fig. 2.

![Diagram of sectioning process](image)

**Fig. 2. Steps in the sectioning method**

3.1. PREPARATION OF TEST SPECIMEN

3.1.1. LOCATION OF SPECIMEN

The location of the test piece along the length of the material is the first step to be performed. To reduce end effects, the test section must be far enough from the ends.

A distance 1.5 to 2.0 times lateral dimension is recommended [1], though theoretically a ratio of 1.0 is sufficient.

3.1.2. PREPARATION OF GAGE HOLES

It is important to prepare the gage holes with care since the accuracy in the res-
tings depends mainly on the type of gage holes. The hole and gage point details for the specimen used in this work are shown in Fig. 3.

![Figure 3: Detail of gage point and gage hole](image)

3.1.3. SECTIONING LOCATIONS

The number of longitudinal strips to be cut depends on the variation of the residual stresses. Steep gradients in residual stresses, for example, would require closer spacings for longitudinal cuttings. To determine residual stresses with a lower number of longitudinal cuts, the method of partial sectioning may be utilized (see Fig. 1). This method requires a prior knowledge of the pattern of residual stress distribution. In order to make proper cutting locations, a fair estimate of the pattern rather than the magnitude of residual stresses would be of importance. The location of a cut for partial sectioning is so determined that it lies near or at the transitions of residual stress gradients.

3.2. MEASURING TECHNIQUE

3.2.1. DIGITAL ELECTRONIC MEASURING MACHINE

The digital electronic measuring machine is used as an instrument for marking

![Figure 4: Digital electronic measuring machine, 30](image)
the gage length and measuring it before the processing and after the sectioning. The machine, as shown in Fig. 4, is three-axis measuring and a new approach to the problem of measuring. By using a coordinate measuring machine, the part is stationary and the machine moves around it in three axes. The result is more accuracy and speed comparing with other measuring instruments. /2/.

3.2.2. PROCEDURE OF MEASUREMENT

In order to avoid the curvature due to processing or sectioning, the gage length is taken 100 mm. When the curvature is large, correction must be made to this strain computation. On a curved strip, the measurement that can be taken with ease in the offset of the arc over the gage length. Using the offset and the change in chord length as the measurement quantities, the true strain may be approximated as:

$$\bar{e} = \frac{\Delta L}{L} + \frac{b/L}{6(b/L)^2 + 1}$$  \hspace{1cm} (1)

where

\(\Delta L/L\) = strain measured by extensometer

\(b/L\) = ratio of offset to gage length

For strain measurements, the following steps are followed.

1. The contact point (probe tip) is inserted into drilled hole.
2. Each measuring unit is controlled into zero set.
3. Motion between the two points of the gage length is measured directly by digital readouts of measuring head position in each axis.

First of all, attention should be given to the importance of obtaining a good set of initial readings, since these cannot be duplicated after the specimen has been cut. Some relevant items to be taken into consideration are:

- Cleaning the gage hole using cleaning solution and air blast before taking any measurement.
- Taking three sets of measurements for each gage length unless great variation persists, in which case making a new set of holes would be advised.
- Protecting the gage holes from damage (such as by covering with tape) which may occur during moving, handling, sawing, etc.

3.3. EVALUATION OF DATA

The computation of residual stresses from measured strains is based on the assumption that the dimensional changes caused by the relaxation are purely linear elastic.
Since strains are read at top and bottom surfaces, evaluation of residual stresses at the respective surface are computed using experimental data.

Let $\bar{E}$ be the average value of the readings taken on one gage length. For each gage length,

$$\bar{E} = \frac{1}{n} \sum_{j=1}^{n} E_j$$

where

- $n$ is the number of readings for one gage length,
- $E_j$ is the measured value for each cycle.

Let $L_i'$ be the initial measured gage length and $L_f'$ the final measured gage length.

Then the total strain due to relaxation is

$$E_i = \frac{L_i - L_f}{L}$$

Using Hooke's law the residual stress at the measured surface is

$$\sigma = -\frac{E_i E}{E_y}$$

By virtue of the linear strain distribution postulated in the beam theory, the average axial stress $\sigma$ in terms of top and bottom measured strains $E_t$ and $E_b$ is

$$\sigma = \frac{E_t E_b}{E_o}$$

"E" is Young's modulus.

3.3.1 EXPERIMENTAL WORK

Flat bars and $\Gamma-$ cross-sections with the dimensions shown in Fig. 5, were selected for experimental investigations. Also, the dimensioning for gage hole and cutting location used on the flat bar and $\Gamma-$ cross section test piece are shown in Fig. 5a. The total number of cuts for sectioning are 10,15 for flat bar and $\Gamma-$ cross-section respectively.

All the investigated specimens were stress-relieved by annealing to omit the influence of residual stresses due to the production processes of steel.

To ensure that the specimens are free from the internal forces and residual stresses which may existed during the production processes of steel, before welding, all the specimens are heated to 620°C and then slowly cooled to the room temperature. A sketch of heating and cooling diagram of the specimen is shown in Fig. 6.

Residual stresses formed due to sawing alone depend, among many other factors, on the spacing of the saw cuts, the plate thickness and the speed of sawing. For one particular set of parameters, the local stress at the saw-cut edge was observed to be the order of 5 to 15 MPa in compression (measured by Blind-Hall Drilling method). Accordingly, the effect of sawing is very small and can be neglected.

A sketch of sawing machine is shown in Fig. 7.
a) Gage hole location and sectioning detail

$\gamma$-section (200, 100, 12)  Flatt bar (200, 14)

(tensile strength: 400 MPa and yield point: 310 MPa)

b) Locations of strain gage rosettes on the surface of the specimen

Fig. 5. Dimensions of the elements used in the experimental work.
Fig. 6. Sketch of heating and cooling diagram

Fig. 7. Sawing machine.

For comparing the results of residual stresses measured by sectioning method, the Blind-Hole Drilling method is used.

With the blind hole drilling method, after installing strain sensors (rosette 1.5/120 RM 61) on the part surface, a small, shallow hole is drilled in the surface. In Ref. 4, a detailed description is available on the measuring procedure of this method. After drilling, the change in strain in the immediate vicinity of the hole is measured, and the relaxed residual stresses are computed from these data.

To determine the values of the principal residual stresses in a particular part, the strain relaxation values $\varepsilon_1$, $\varepsilon_2$, and $\varepsilon_3$ are substituted in the following formula,

$$\sigma = \frac{1}{\mu} (\Delta \varepsilon_1 + \Delta \varepsilon_2) \pm \frac{E}{4\mu} \sqrt{(\Delta \varepsilon_1 + \Delta \varepsilon_2 - 2\Delta \varepsilon_3)^2 + (\Delta \varepsilon_2 - \Delta \varepsilon_3)^2}$$

(6)
where
\[ E = \text{Elastic Modulus} \]
\[ A, B = \text{Constants} \]
The constant "A", "B" can be determined as follows:
\[
A = a^2(1+v) \frac{1}{2r_1} \]
\[
B = \frac{2a^2}{\varepsilon_0 r_1} \left[ 1 - a^2(1+v) \frac{r_0^2 + r_1^2 + r_2^2}{4r_1^2} \right] \]

(7)

Fig. 7, showing the out-side, in-side and bore radius on the rosette 1.5/120 RY 61.

where
\[ v = \text{Poisson's ratio} \]
\[ r_0 = \text{Outer radius of strain gage (rosette)} \]
\[ r_1 = \text{Inner radius of strain gage (rosette)} \]
\[ a = \text{Radius of the (bore)drilling hole} \]

4. EXPERIMENTAL RESULTS AND DISCUSSION

Comparison between the residual stresses distribution for two different sections (flat bar and T-section) conjunction with plating by welding as measured by the sectional and blind-hole drilling methods, is shown in Fig. 9.

Generally, the shape of residual stresses distribution measured by sectional and blind-hole drilling methods is the same. There is a little difference between the results obtained by sectional and blind-hole drilling methods.

The differences in the stresses distribution of the investigations were probably caused by the following.
Fig. 9. Residual stress distribution in flat bar and T-section in conjunction with peening.

(-- Sectional method, --- Blind-hole drilling method)
Sectional method:
- Cutting operation: Temperature changes appear to be the major cause of errors introduced during residual stresses measurement.
- Errors in measured strain which caused with curvature due to processing or sectioning are reflected in the residual stresses measurement.

Blind-hole drilling method:
- The hole acts as a stress raiser and, provided the stress magnitude is sufficient, localized plasticity will occur around the hole which leads to additional error in the predicted stresses. This error increases 10%, /9/.
- Large error in the predicted principal stresses can result from small error in measured strain.
- Mechanical cutting methods currently used for hole forming introduce cutting stresses which are reflected in the strain measurements.

To obtain satisfactory results with the sectioning method it is important to perform a careful preparation of the test piece such as proper location of the test section, gage hole location and cutting position. Preparation of gage holes must be performed with care since unreliable readings can result if the holes are not prepared in a proper manner.

5. CONCLUSIONS

The following conclusions may be made for the residual stresses measurement:
1. The method of sectioning is adequate, accurate and economical for the residual stresses measurement in structural elements of ships. By increasing number of sections, accuracy of results can be increased.
2. Digital electronic automation machine can be used as an instrument for marking and measuring gage length from which strain and stress can be determined. By using this machine the sectional method is more simple and economical than other methods of measuring residual stresses.
3. As the disadvantages of the sectioning method, it is considered destructive method, gives average stresses over the area of the piece removed from the specimen, not suitable for measuring locally concentrated stresses for a whole part. Panel or other scaled models the method is expensive and time consuming.

6. REFERENCES

/2/ DEA: Optional Electronics for TOL, Gamma, Beta and Sigma. Measuring Centers for 3D Inspection and Layout, Torino, Italy.